

### IPE300-H=488 Properties :

$$b = 0.15 \text{ m} \quad t_f = 0.0107 \text{ m}$$

$$h = 0.4875 \text{ m} \quad t_w = 0.0071 \text{ m}$$

$$A = 0.00661437 \text{ m}^2$$

$$I_x = 0.00024744 \text{ m}^4$$

$$I_y = 6.0364\text{E-}06 \text{ m}^4$$

$$S_x = 8.04854\text{E-}05 \text{ m}^3$$

$$S_y = 8.04854\text{E-}05 \text{ m}^3$$

$$Z_x = 0.00117278 \text{ m}^3$$

$$Z_y = 0.00012681 \text{ m}^3$$

$$i_x = 0.193415 \text{ m}$$

$$i_y = 0.0302096 \text{ m}$$

$$C_w = 3.43076\text{E-}07 \text{ m}^6$$

$$J_c = 2.23554\text{E-}07 \text{ m}^4$$

### Material Properties: S275

$$F_y \quad : \text{Yield stress} \quad = 275000 \text{ kN/m}^2$$

$$F_u \quad : \text{Ultimate stres} \quad = 430000 \text{ kN/m}^2$$

$$E \quad : \text{Elasticity modulus} \quad = 2.000\text{E+}8 \text{ kN/m}^2$$

$$G \quad : \text{Shear modulus} \quad = 7.692\text{E+}7 \text{ kN/m}^2$$

### **The psi value of the moment diagram, as per Table 18**

psi: 0.838901135592712

### **The c1, c2 and c3 constants as per Table 42 (Annex E), used in Clause E-1.2**

Bending Moment Diagram Type: Type1

c1: 1.000

c2: 0.000

c3: 1.000

### **Calculation of the Elastic Critical Moment as per Annex E**

$$M_{cr} = c_1 \frac{\pi^2 EI_y}{L_{IT}} \left\{ \left[ \left( \frac{K}{K_w} \right)^2 \frac{I_w}{I_y} + \frac{GI_t (L_{IT})^2}{\pi^2 EI_y} + (c_2 y_g - c_3 y_j) \right]^{0.5} - (c_2 y_g - c_3 y_j) \right\}$$

$$c_1 \frac{\pi^2 EI_y}{L_{IT}} = 1 \frac{\pi^2 \cdot 2.000E+8 \cdot 6.04E-06}{1.72} = 6935.77$$

$$\left( \frac{K}{K_w} \right)^2 \frac{I_w}{I_y} = \left( \frac{1}{1} \right)^2 \frac{3E-07}{6.04E-06} = 0.06$$

$$\frac{GI_t (L_{IT})^2}{\pi^2 EI_y} = \frac{7.692E+7 \cdot 2.24E-07 \cdot (1.72)^2}{\pi^2 \cdot 2.000E+8 \cdot 6.04E-06} = 0.00426$$

$$(c_2 y_g - c_3 y_j) = 0.0 \cdot 0.0 - 1 \cdot 0.0 = 0.0$$

$$M_{cr} = 6935.77 \{ [0.06 + 0.00426 + 0.0]^{0.5} - 0.0 \}$$

$$M_{cr} = 1714.33 \text{ kN.m}$$

Check against flexure and axial force

### Load Combinations

ComboType= ULS

Combo Name= 1.5xD+1.5xSELF+1.5xPUR+1.5xSHE+1.5xSNOW1

Station

Loc : 0.859 m  
N : -107.74 kN  
M33 : -209.18 m.kN  
V22 : 0.0 kN  
M22 : 0.0 m.kN  
V33 : 80.8 kN  
T : 0.0 kN

### **Classifies a H Section bending about ZZ axis**

Type of extreme section fiber= Top

$$\sigma = \frac{P}{A} + \frac{M_3}{W_{ely}} = \frac{-107.74}{0.00661437249999999} + \frac{-209.18}{0.00102} = -222353.25 \text{ kN/m}^2$$

$-222353.25 < 0 \Rightarrow \text{Compression}$

Rolled Section flange

$$\text{ratio} \leq 9.4 \cdot \varepsilon = 7.01 \leq 9.4 \cdot 0.953 \Rightarrow 1$$

Type of extreme section fiber= Top

$$\sigma = \frac{P}{A} + \frac{M_3}{W_{ely}} = \frac{-107.74}{0.00661437249999999} + \frac{-209.18}{0.00102} = -222353.25 \text{ kN/m}^2$$

$-222353.25 < 0 \Rightarrow \text{Compression}$

$$d = h - t_f = 0.4875 - 0.0107 = 0.466$$

$$r_1 = \frac{\frac{N}{ratio \cdot Area}}{d \cdot t_w \cdot f_y \cdot Y_0} = \frac{\frac{107.74}{1}}{0.466 \cdot 0.0071 \cdot 275000 \cdot 1.1} = 0.13$$

$$r_2 = \frac{\frac{N}{ratio \cdot Area}}{Area \cdot f_y \cdot Y_0} = \frac{\frac{107.74}{1}}{0.00661 \cdot 275000 \cdot 1.1} = 0.0652$$

### Classifies a H Section bending about YY axis

Type of extreme section fiber = Left

$$\sigma = \frac{P}{A} - \frac{M_2}{W_{elz}} = \frac{-107.74}{0.00661437249999999} - \frac{0.0}{8.05E-05} = -16289.06 \text{ kN/m}^2$$

$-16289.06 < 0 \Rightarrow \text{Compression}$

Type of extreme section fiber = Right

$$\sigma = \frac{P}{A} + \frac{M_2}{W_{elz}} = \frac{-107.74}{0.00661437249999999} + \frac{0.0}{8.05E-05} = -16289.06 \text{ kN/m}^2$$

$-16289.06 < 0 \Rightarrow \text{Compression}$

Rolled Section flange

$$ratio \leq 9.4 \cdot \varepsilon = 7.01 \leq 9.4 \cdot 0.953 \Rightarrow 1$$

$$d = h - t_f = 0.4875 - 0.0107 = 0.466$$

$$r_1 = \frac{\frac{N}{ratio \cdot Area}}{d \cdot t_w \cdot f_y \cdot Y_0} = \frac{\frac{107.74}{1}}{0.466 \cdot 0.0071 \cdot 275000 \cdot 1.1} = 0.13$$

$$r_2 = \frac{\frac{N}{ratio \cdot Area}}{Area \cdot f_y \cdot Y_0} = \frac{\frac{107.74}{1}}{0.00661 \cdot 275000 \cdot 1.1} = 0.0652$$

### Checks whether the calculated effective slenderness ratio is within the allowable limits of Table 3

MemberType: No1      max(KL/r)=180

(A member carrying compressive loads resulting from dead loads and imposed loads)

$$\left(\frac{KL}{r}\right)_{z-z} = \frac{k_z L_z}{i_z} = \frac{k_z L_z}{\sqrt{\frac{I_z}{A_g}}} = \frac{1 \cdot 1.72}{\sqrt{\frac{6.04E-06}{0.00661}}} = 8.88 \leq 180 \text{ Satisfactory } \checkmark \text{ o.k.}$$

$$\left(\frac{KL}{r}\right)_{y-y} = \frac{k_y L_y}{i_y} = \frac{k_y L_y}{\sqrt{\frac{I_y}{A_g}}} = \frac{1 \cdot 1.72}{\sqrt{\frac{0.000247}{0.00661}}} = 56.87 \leq 180 \text{ Satisfactory } \checkmark \text{ o.k.}$$

### COMPRESSION CHECK AS PER SECTION 7

### - About Z-Z

Buckling Class (IPE300-H=488) – Table10 : 1

$\alpha$  from Table 7 (1) = 0.21

Euler buckling stress ( $f_{cc}$ ) from Section 7.1.2.1

$$f_{cc} = \frac{\pi^2 E}{(KL/r)_{z-z}^2} = \frac{\pi^2 2.000E+8}{(8.88)^2} = 2.502E+7$$

$$\lambda = \sqrt{\frac{f_y}{f_{cc}}} = \sqrt{\frac{275000}{2.502E+7}} = 0.105$$

$$\Phi = 0.5[1 + \alpha(\lambda - 0.2) + \lambda^2] = 0.5[1 + 0.21(0.105 - 0.2) + 0.105^2] = 0.496$$

$$\chi = \frac{1}{[\Phi + (\Phi^2 - \lambda^2)^{0.5}]} = \frac{1}{[0.496 + (0.496^2 - 0.105^2)^{0.5}]} = 1.02$$

$$f_{cd} = \min\{\chi f_y / \gamma_{m0}, f_{cc} / \gamma_{m0}\} = \min\{1.02 \cdot 275000 / 1.1, 2.502E+7 / 1.1\} = 250000 \text{ kN/m}^2$$

$$P_d = A_e \cdot f_{cd} = 0.00661 \cdot 250000 = 1653.59 \text{ kN}$$

$$\text{Ratio}_{z-z} = \frac{P}{P_d} = \frac{107.74}{1653.59} = 0.0652 \leq 1.0 \Rightarrow \text{Satisfactory} \checkmark \text{ o.k.}$$

### - About Y-Y

Buckling Class (IPE300-H=488) – Table10 : 2

$\alpha$  from Table 7 (2) = 0.34

Euler buckling stress ( $f_{cc}$ ) from Section 7.1.2.1

$$f_{cc} = \frac{\pi^2 E}{(KL/r)_{y-y}^2} = \frac{\pi^2 2.000E+8}{(56.9)^2} = 610369.61$$

$$\lambda = \sqrt{\frac{f_y}{f_{cc}}} = \sqrt{\frac{275000}{610369.61}} = 0.671$$

$$\Phi = 0.5[1 + \alpha(\lambda - 0.2) + \lambda^2] = 0.5[1 + 0.34(0.671 - 0.2) + 0.671^2] = 0.805$$

$$\chi = \frac{1}{[\Phi + (\Phi^2 - \lambda^2)^{0.5}]} = \frac{1}{[0.805 + (0.805^2 - 0.671^2)^{0.5}]} = 0.8$$

$$f_{cd} = \min\{\chi f_y / \gamma_{m0}, f_{cc} / \gamma_{m0}\} = \min\{0.8 \cdot 275000 / 1.1, 610369.61 / 1.1\} = 199926.57 \text{ kN/m}^2$$

$$P_d = A_e \cdot f_{cd} = 0.00661 \cdot 199926.57 = 1327.59 \text{ kN}$$

$$\text{Ratio}_{y-y} = \frac{P}{P_d} = \frac{107.74}{1327.59} = 0.0811 \leq 1.0 \Rightarrow \text{Satisfactory} \checkmark \text{ o.k.}$$

### Vn: SHEAR RESISTANCE IN Y DIRECTION, AS PER 8.4.1

Section Name= IPE300-H=488

web Depth (d)= 0.4661 m

web Thickness= 0.0071 m

$$V_{p,y} = \frac{A_v \cdot F_{yw}}{\sqrt{3}} = \frac{0.0038 \cdot 275000}{\sqrt{3}} = 603.54 \text{ kN}$$

Transverse Stiffeners Provided Only At Supports

$$K_{w,y} = 5.35$$

**Resistance to shear buckling shall be verified, as per 8.4.2.1**

Frame Element Web Has No Stiffeners

$$\varepsilon = \sqrt{250000/f_y} = \sqrt{250000/275000} = 0.953$$

$$\frac{d}{t_w} = \frac{0.466}{0.0071} = 65.6 > 67 \varepsilon \Rightarrow \text{Consider}$$

**Non-dimensional web slenderness ratio for buckling stress**

$$\tau_{cr,y} = \frac{K_{w,y} \cdot \pi^2 \cdot E}{12(1-\mu^2)(d/t_w)^2} = \frac{5.35 \cdot \pi^2 \cdot 2.000E+8}{12(1-s\mu^2)(0.466/0.0071)^2} = 224398.2$$

$$\lambda_{w,y} = \sqrt{f_{yw}(\sqrt{3}\tau_{cr,y})} / = \sqrt{s f_{yw}(\sqrt{3} \cdot 224398.2)} / = 5.35$$

**Shear stress corresponding to web buckling**

$$\lambda_{w,y} > 1.2 \Rightarrow \tau_{b,y} = f_{yw}(\sqrt{3} \cdot \lambda_{w,y}^2) / = 275000 \sqrt{3} \cdot 0.841 = 153543.86/$$

$$V_{cr,y} = A_w \cdot \tau_{b,y} = s A_w \cdot 153543.86 = 583.67$$

**Design shear strength in Y direction as per 8.4.1 and 8.4.2**

$$V_n = \min\{V_p, V_{cr}\} = \min\{s V_p, s V_{cr}\} = 583.67 \text{ kN}$$

**Design shear ratio in Y direction**

$$Shear_{ratio,y} = \frac{V}{V_n} = \frac{80.8}{530.61} = 0.152 \leq 1.0 \Rightarrow \text{Satisfactory} \checkmark \text{ o.k.}$$

**Design bending strength of the section (Mdz) 8.2.1.2**

**Section class for flexure about ZZ axis: 4**

$$\beta_b = \frac{W_{el,y}}{W_{pl,y}} = 0.866 \text{ for semi compact sections}$$

$$d / t_w > 67 \Rightarrow \varepsilon$$

$$M_{d,y} = \beta_b \cdot W_{pl,y} \cdot f_{y/YM}$$

$$M_{d,y} = 0.866 \cdot 0.000765 \cdot 275000 / 1.1 = 165.6 \text{ kN.m}$$

**Bending ration about ZZ axis**

$$Bending_{ratio,z} = M_{3,3} / M_{d,z} = M_{dv,z} = -209.18 / 165.6 = 1.26 > 1.0 \text{ Not satisfactory } \times$$

**Calculation of the resistance against Lateral Torsional Buckling (MdLTB) 8.2.2**

$\alpha_{LT} = 0.21$  for rolled steel sections

$$\lambda_{LT} = \min \left\{ \sqrt{\beta_b \cdot W_{pl,y} \cdot f_y / M_{cr}}, \sqrt{1.2 \cdot W_{el,y} \cdot f_y / M_{cr}} \right\}$$

$$\lambda_{LT} = \min \left\{ \sqrt{0.866 \cdot 0.00117 \cdot 275000 / s M_{cr}}, \sqrt{1.2 \cdot 0.00102 \cdot 275000 / 1714.33} \right\} = 0.404$$

$$\Phi_{LT} = 0.5 [1 + \alpha_{LT} \cdot (\lambda_{LT} - 0.2) + \lambda_{LT}^2] = 0.5 [1 + 0.21 \cdot (0.404 - 0.2) + 0.404^2] = 0.603$$

$$\chi_{LT} = \min \left\{ \frac{1}{\{\Phi_{LT} [\Phi_{LT}^2 - \lambda_{LT}^2]\}^{0.5}}, 1 \right\} = \min \left\{ \frac{1}{\{0.603 [0.603^2 - 0.404^2]\}^{0.5}}, 1 \right\} = 0.952$$

$$f_{bd} = \chi_{LT} \cdot f_y / \gamma_{M0} = 0.952 \cdot 275000 / 1.1 = 237963.13 \text{ kN/m}^2$$

$$M_{d,LTB} = \beta_b \cdot W_{pl,y} \cdot f_{bd} = 0.866 \cdot 0.00117 \cdot 237963.13 = 241.57 \text{ kN.m}$$

$$LTB_{ratio,z} = M_{3.3} / M_{LTB,z} = -209.18 / 241.57 = 0.866 \leq 1.0 \text{ Satisfactory } \checkmark \text{ o.k.}$$

### Vn: SHEAR RESISTANCE IN Z DIRECTION, AS PER 8.4.1

Section Name= IPE300-H=488

web Depth (d)= 0.15 m

web Thickness= 0.0107 m

$$V_{p,z} = \frac{A_v \cdot F_{yw}}{\sqrt{3}} = \frac{0.00331 \cdot 275000}{\sqrt{3}} = 0.0 \text{ kN}$$

### Resistance to shear buckling shall be verified, as per 8.4.2.1

Frame Element Web Has No Stiffeners

$$\varepsilon = \sqrt{250000 / f_y} = \sqrt{250000 / 275000} = 0.953$$

$$\frac{d}{t_w} = \frac{0.15}{0.0107} = 14 \leq 67 \varepsilon \Rightarrow \text{Not Consider}$$

### Design shear ratio in Z direction

$$Shear_{ratio,z} = \frac{V}{V_n} = \frac{80.8}{477.04} = 0.169 \leq 1.0 \Rightarrow \text{Satisfactory } \checkmark \text{ o.k.}$$

### Combined Axial force and Bending moment check regarding overall member strength, as per 9.3.2

$$n_y = P / P_{dy} = -107.74 / 1322.39 = 0.0815$$

$$n_z = P / P_{dz} = -107.74 / 1653.59 = 0.0652$$

$$K_y = \min \{ 1 + (\lambda_y - 0.2) \cdot n_y, 1 + 0.8 \cdot n_y \} = \min \{ 1 + (0.671 - 0.2) \cdot 0.0815, 1 + 0.8 \cdot 0.0815 \} = 1.04$$

$$K_z = \min \{ 1 + (\lambda_z - 0.2) \cdot n_z, 1 + 0.8 \cdot n_z \} = \min \{ 1 + (0.105 - 0.2) \cdot 0.0652, 1 + 0.8 \cdot 0.0652 \} = 0.994$$

$$K_{LT} = \max \left\{ 1 - \frac{0.1 \cdot \lambda_{LT} \cdot n_y}{C_{mLT} - 0.25}, 1 - \frac{0.1 \cdot n_y}{C_{mLT} - 0.25} \right\} = \max \left\{ 1 - \frac{0.1 \cdot 0.404 \cdot 0.0815}{0.6 - 0.25}, 1 - \frac{0.1 \cdot 0.0815}{0.6 - 0.25} \right\} = 0.994$$



$$\frac{P}{P_{dy}} + K_y \frac{C_{my} \cdot M_y}{M_{dy}} + K_{LT} \cdot \frac{M_z}{M_{dz}} \leq 1.0$$

$$\frac{-107.74}{1322.39} + 1.04 \frac{0.6 \cdot 0.0}{20.1} + 0.991 \cdot \frac{-209.18}{165.6} = 1.33 > \text{Not satisfactory } \times$$

$$\frac{P}{P_{dz}} + 0.6 \cdot K_y \frac{C_{my} \cdot M_y}{M_{dy}} + K_z \cdot \frac{C_{mz} \cdot M_z}{M_{dz}} \leq 1.0$$

$$\frac{-107.74}{1653.59} + 0.6 \cdot 1.04 \frac{0.6 \cdot 0.0}{20.1} + 0.994 \cdot \frac{0.6 \cdot -209.18}{165.6} = 0.818 > \text{Satisfactory } \checkmark \text{ o.k.}$$

### Combined Axial force and Bending moment check regarding section strength, as per 9.3.1

In the design of members subjected to combined axial and bending moment-Conservatively

$$\text{Section Strength ratio} = \frac{N}{N_d} + \frac{M_y}{M_{ndy}} + \frac{M_z}{M_{ndz}} \leq 1.0$$

$$\text{Section Strength ratio} = \frac{107.74}{1653.59} + \frac{0.0}{20.1} + \frac{209.18}{165.6} > 1.33 \text{ Not satisfactory } \times$$